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GSE FOR BALLOON-BORNE I.M.S.: DECOMMUTATOR AND D/A UNITS

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Northeastern University Electronics Research Laboratory Boston, Massachusetts 02115

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and/or 8 bit words. Communications interface for commands and messages to and from the airborne instrumentation was also provided. The second unit converted the digital data, supplied by the computer, into analog signals for display and recording. An IEEE-488 compatible interface was used for data transfer.						

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TABLE OF SELECTED ACRONYMS AND ABBREVIATIONS

1.	ATN	ATTENTION
2.	CMOS	COMPLIMENTARY METAL OXIDE SEMICONDUCTOR
3.	CPU	CENTRAL PROCESSING UNIT
4.	CRT	CATHODE RAY TUBE
5.	cs	CHIP SELECT
6.	D/A	DIGITAL TO ANALOG
7.	DAV	DATA VALID
8.	D to A	DIGITAL TO ANALOG
9.	E0I	END OR IDENTIFY
0.	EPROM	ELECTRICALLY PROGRAMMABLE READ ONLY MEMORY
۱.	FIFO	FIRST IN, FIRST OUT
2.	HEXFET	POWER METAL OXIDE SEMICONDUCTOR (Trade name of
		International Rectifier)
3.	IFC	INTERFACE CLEAR
4.	LED	LIGHT EMITTING DIODE
5.	LSB	LEAST SIGNIFICANT BIT
6.	μΡ	MICROPROCESSOR
7.	NDAC	NOT DATA ACCEPTED
8.	NRFD	NOT READY FOR DATA
9.	PCM	PULSE-CODE MODULATION
0.	RAM	RANDOM ACCESS MEMORY
1.	RTS	REQUEST TO SEND DATA
22.	SRQ	SERVICE REDUEST
3.	TTL	TRANSISTOR-TRANSISTOR LOGIC
24.	USART	UNIVERSAL SYNCHRONOUS/ASYNCHRONOUS RECEIVER/

TRANSMITTER

INTRODUCTION

Two ground based units were developed to support a Hewlett Packard HP9845C system in data processing operations during the flight of a Balloon Borne Ion Mass Spectrometer (BBIMS). One of the support units was designated to decommutate the incoming PCM data, while the other was developed to convert digital data, supplied by the computer, into analog signals for display.

The data gathering modes of the mass spectrometer could be controlled from the ground. The instrument could be commanded to execute any one of a number of programs or entire repertoires of programs stored on board. Also, entirely new programs could be transmitted from the ground for execution over a serial command link available for this purpose. Commands and instructions are sent to the mass spectrometer control unit through a command transmitter. A single 8 bit word within the PCM data train was used as the down link for communications from the airborne unit. Detailed description of the BBIMS instrumentation and its capabilities can be found in References 1, 2 and 3. The flexibility of the airborne instrument allowed the ground based scientist to tailor the data gathering process to the existing conditions. This required an almost real time processing and display of data, and the capability to send commands to and to receive communications from the airborne mass spectrometer control unit. Also, a large number of signals were displayed to monitor the status and the performance of the balloon borne instrumentation.

During the first successful flight of the BBIMS instrumentation package in May 1982, the HP9845C system was used to display the monitor

signals and to plot selected portions of the ion spectrum. It received the data from a ground control unit designed primarily to communicate with the airborne control unit. The data was available in a parallel form, one byte at a time, together with a word synchronization pulse. Frame synchronization was not available; therefore, the computer had, for all practical purposes, to decommutate the data. This task consumed a considerable amount of processing time. Commands to the mass spectrometer had to be sent through the ground control unit.

The newly developed system overcomes many of the disadvantages of the earlier ones and is shown in the block diagram of Figure 1.

The decommutator, one of the newly constructed ground support units, relieved the HP9845C system from the task of the PCM data decommutation. It presented the selected data, at the convenience of the computer, through the IEEE-488 bus. Also, it provided a communications interface between the computer and the serial command link and relayed the messages from the balloon-borne mass spectrometer control unit, within the PCM data stream, to the HP computer. Provisions were included for the use of a printer to obtain a hard copy of all communications.

The D/A UNIT was designed to convert eight bytes of data into analog signals. Microammeters and output terminals were provided for the display and for the recording of the analog data. The data channels were selected by the computer.

This report describes the capabilities and the operation of these two support units. Detailed descriptions of the circuits are provided.

I. DECOMMUTATOR

The primary task of the BBIMS-2 Decommutator Unit was, as its name implies, to decommutate the incoming PCM data from the balloon borne instruments. The decommutated data was made available to the user through an IEEE-488 interface bus and also in a parallel form as 16 and/or 8 bit words. The parallel data was accompanied by appropriate synchronization pulses. The secondary task of the unit was to provide communications/command interface to the airborne mass spectrometer control unit. All communications passing through the interface were available at yet another port for use by a printer or a CRT terminal. The communications ports were compatible with the **RS**-232-C⁵ link. LED indicators were provided to monitor the operation of the decommutator unit.

A. OPERATION

The incoming PCM data from the balloon borne instruments was formated into a 20 word frame transmitted at 12 kilobits per second. Eight bit words were used. A 16 bit pattern (EB90H) provided the frame synchronization. The subframe consisted of 96 data words. Two of the subframe words appeared in each frame. Those words were preceded by a subframe identification code. Most of the other words within the frame carried data associated with the quadrupole ion mass spectrometer, while the subframe was mostly used for monitor data or for the data from supporting experiments. A notable exception were two words within the frame. One contained the ONE's count of the frame, but did not include the frame synchronization pattern nor the ONE's in the count word itself. The count was intended to

show the number of non-canceling errors within the frame and, therefore, provided a rough indication of the quality of the telemetry link. The other word carried communications from the mass spectrometer flight control unit to the ground station. The communications words were inserted into the PCM data stream at an equivalent rate of 300 baud. This rate was chosen to accommodate the relatively slow ground support equipment. Also, since the data gathering rate of the quadrupole ion mass spectrometer was quite low, not every frame of the PCM pulse stream carried new and/or valid data. The first word of a frame was used to indicate the status of the data and the presence or absence of a new communications word within that frame. The decommutation process was controlled, to a large extent, by the status of two bits within that word.

Whenever the designated bit (MSB) in the first frame word indicated that a new and valid data was present in the frame, the data was extracted and stored. Only selected words were thus processed. The selection was determined by software. Primarily, the words directly associated with the data gathered by the ion mass spectrometer were chosen.

Eight 32 word registers were created for storage of that data in the 1K RAM included in the decommutator circuits. Each of the registers was dedicated to one minor frame. Thus eight minor frames of data could be stored consecutively before the very first frame of data was lost due to an overwrite. Although only 13 words needed storage locations, the registers were purposely extended to 32 locations. The balloon borne PCM encoder format was software programmable; therefore, future flights can utilize a different format if many more

words within a frame require storage in the RAM. To maintain this flexibility the very first location of each storage register was reserved for the number of words actually stored; therefore, only the locations used for storage were addressed during the transfer of the data to the peripheral processing equipment. Once the data was read, ZEROS were placed into the first location of the storage anster indicating that the register has been processed.

Subframe data, which contained mostly the monitor sich of the airborne instrumentation was extracted and stored continuously. For that purpose two 256 byte registers were allocated in the RAM. Once again, the registers were able to accommodate consideratly more data than was available in each subframe; therefore, future subframes may contain more data. The subframe data words were stored consecutively. Once all of the subframe words were stored, a binary number representing the number of words contained in the register was placed into the first location. During the following major frame of the PCM data the second storage register was filled. By that time, hopefully, the data from the first storage register were transferred to the peripheral sources. If not, then an overwrite occurred. The decommutation process was controlled by the subframe identification word which had to precede the first subframe word within a minor frame. A subframe identification word 00H was interpreted as a beginning of the subframe.

The data stored in the RAM was available only through the IEEE-488 interface. A subroutine searched for a full storage register containing the mass spectrometer data. Once found, the decommutator established contact with the bus controller and then transmitted the

stored data. When transfer was a supreted, the program returned to check if the next consecutive register containing the mass spectrometer data was full. Thus the data transmission was carried out on FIFO basis. Only when all eight of the frame storage data registers were empty, the contents of a full subframe data storage register were output. Therefore, the mass spectrometer data had a priority over the subframe data. To differentiate between the two data types, an identification code, six for the subframe data, seven for the minor frame data, was transmitted as the first word. The word count stored in the first location of the storage register was not transmitted. The end of the data transfer was marked by a TRUE state on the EOI line during the transmission of the last data word. Provisions were included to continue data transmission from the same storage register following an interruption by IFC command from the controller. When the frame synchronization was lost the data processing was suspended.

The parallel data was available in real time. It remained stable for the duration of approximately one PCM data word. The 16 bit words were presented with the odd numbered PCM data word as the most significant byte. A word sync pulse lasting one-half of the duration of a PCM word and centered in that time period marked the availability of the 16 bit word. The byte wide data was also marked with a similar pulse on a different output line. Finally, when the whole synchronization word (16 bits) was available at the parallel data port, a synchronization pulse coinciding with and of the same duration as the second frame synchronization byte (90H) appeared on the third output line. The parallel data, although its integrity could be questionable, was available even when the frame synchronization was lost.

The communications from the airborne mass spectrometer control unit to the ground station were processed in somewhat similar manner as the PCM data. When a selected bit (3rd. USB) in the first word of the PCM data train indicated that a new communications word was present in the frame, that word was extracted and stored. A 256 byte register was allocated in the RAM for that purpose. Next, the status of the decommutator USART, used to retransmit the communications from the flight control unit to the xontroller (HP9845C), was checked. When it was found to be empty, the previously stored word or, it none were available, the very recent word was transferred into the USART. When no new communications were present in the frame, then only the USART was serviced during the time interval allocated for the communications word within the frame.

B. PROGRAM

The program controlling the operations of the decommutator unit could be subdivided into three groups of subroutines each performing distinct tasks. The system was prepared for operation by the initialization routines. Data transfer and handshakes with the bus controller were handled by another set of subroutines. The decommutation and storage process was controlled by a subprogram designed to accommodate the frame format. This latter program called upon appropriate subroutines to perform the necessary decommutation and storage tasks. The decommutation subprogram was entered from the data transfer subroutines upon an interrupt generated by the hardware. Most of the time the system operated in the data transfer group of subroutines.

Upon TURN-ON or RESET the program initialized the decommutator ports, established clock frequency and continued into a subroutine

which determined the polarity of the incoming PCM data. Signal contitioning circuits were set to present the data in the proper polarity to the decommutator shift registers. Initial frame synchronization was also established at this point. From there, the system entered a routine where a search was conducted for a full decommutated data storage register. When a full register was found, another routine requested service from the controller of the IEEE-488 bus. After the designation to be a TALKER (code 29D) has been received, another subroutine transferred data from the storage register to the bus. When the transfer of the data from that register has been completed the system returned to the routine which searched for another full storage register.

The tasks of the data transfer from the storage registers in the RAM to the bus were interrupted whenever a PCM data word was ready for processing. Upon interruption, the contents of all CPU registers were saved. The control was transferred to the decommutator subprogram which selected an appropriate subroutine to process that particular word within the PCM data frame. To adapt the decommutator to a different PCM data format only the decommutator subprogram had to be changed. The subprogram consisted mainly of a series of jump instructions transferring control to the various processing routines. Upon completion of a given task, the program restored the contents of the CPU registers and returned to continue the interrupted data transfer process. The subroutines of the program and their flow graphs are presented in the APPENDIX at the end of this report.

C. CIRCUITS

The decommutator design was based on the #085 ...P and its family of support components. Some of the support functions were delegated to a sequential logic utilizing CMOS components. The circuit diagram is shown in Figure 2.

The incoming PCM data was transmitted through U20, U25 and the AND-AND-OR gate arrangement (U27) to a 16 bit shift register (U10, U11). These gates were used to invert the PCM data when necessary. Once the program determined, during the initialization process, that an inversion of polarity was required, a negative level was applied to pin 2 of U27 through the port C of U4. This process was non-reversible unless the system was reinitialized.

The clock to enter the data into the serial-in-parallel-out shift register (U15, U16) was derived from the CPU clock through U4 and U24. Synchronization with the incoming PCM data was maintained by resetting U24 through the AND-AND-OR gates (U26) on every transition in the data bit stream (transitions marked by pulses generated by U25) or by utilizing the external clock supplied with the PCM data. When present, the external clock was delayed one quarter of a period by U18 and its period was doubled by U19. After passage through U26 the transitions of this signal caused U25 to generate pulses which were then used to synchronize the internal clock with the incoming data.

The switchover from the data controlled synchronization to the external clock synchronization was accomplished in the gates of U26 in conjunction with the signals generated by U18. This monostable

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detected the presence of the external clock, generated the switchover signals and lit a LED to indicate to the user that indeed the external clock was present.

The frequency of the data shift clock output of U24 was divided by 8 in the bit counter of U28. The output of the counter was used to interrupt the i.P and to latch the data present in the shift register into the parallel data output ports U15 and U16. The synchronization pulses for the user of the parallel data were derived in units 29 and 30 from the signals of U28.

The overall synchronization between the PCM data and the sequential logic circuits was maintained by software. At the beginning of the subroutine which processed the second byte of the frame synchronization word (90H) a pulse from the LSB of port C of U4 reset the bit counter U28. From there on, provided the data and/or the external clock did not have noise induced transitions, the system was synchronized. When the synchronization was lost, it was readily restored when the CPU again detected the PCM frame synchronization word. During the time that the CPU searched for the frame synchronization pattern, additional reset pulses were generated whenever the byte EBH was detected. These pulses were necessary to resynchronize the bit counter in order that the second byte (90H) of the synchronization word could be detected eight bits later. Therefore, the BYTE, the WORD and the FRAME synchronization pulses for the parallel data output were not valid during the search periods.

A LED was used to signal the synchronization status. In absence of signal the light was off. Random signals with an occasional frame synchronization byte pattern produced a flickering light. Synchronization was marked by a steady glow.

Upon receiving an interrupt from the bit counter (U28), the CPU transferred the data byte from the shift register U11 through port A of U5 into the scratch pad RAM of U4. From there, only selected data words were transferred into one of the storage registers in the 1k byte RAM U14.

The communications from the balloon borne ion mass spectrometer control unit to the ground control were transmitted through the USART U17 and the SR-232C interface circuits U21 and U22. When ready to transmit, the USART generated a Request To Send signal ($\overline{\text{RTS}}$). When the request was granted ($\overline{\text{CTS}}$ low), transmission of the messages started (via P1-2) and continued until all of the stored messages were transmitted. The transmission could be interrupted by canceling the permission to send ($\overline{\text{CTS}}$ high). The USART resumed transmission of the interrupted message when the $\overline{\text{CTS}}$ signal was once again received. The end of transmission was indicated when the USART removed the $\overline{\text{RTS}}$ signal from the line. Although the software to receive data through the USART from the command link receiver was not included in the program, hardware connections to the USART were provided for that purpose.

Commands from the ground based control to the mass spectrometer flight control unit arriving on P1-2, were transmitted to the command transmitter via S2-14. The decommutator unit acted as a junction box between the two units and provided an isolated connection to the printer port. The output of the USART was OR'ed with the command signals before entering the port.

The IEEE-488 bus transactions were carried out through U4 and U5. U4 received the signals from the bus which U5 transmitted to

the bus. Port A of U4 monitored the data bus. Port B of the same unit handled the handshakes and the commands. Data transmission took place through port B of U5 in conjunction with the buffers U13. The handshakes were transmitted through port C of U5 and the buffers U12.

Port C of U4 was designated as a general utility port. The synchronization pulse for the bit counter and the signal to control the polarity of the incoming PCM data were transmitted through this port. Also, LED indicators to display the status of the decommutator unit were controlled through that port.

The DAV LED was turned on when the data within the PCM data frame was new and valid. The TTY LED was lit when a communications word was present in the frame. The SR LED indicated that a service request (SRQ) was issued to the controller by the decommutator unit. Finally, the TALK LED turned on when the decommutator started data transmission over the IEEE-488 bus. These LED's were under software control and were turned on for a minimum of 3 PCM data frames.

The chip enable functions were generated by the decoder U2, while U9 latched the low byte of the address for the EPROM U8, where the decommutator program was stored.

II. D/A UNIT

To observe the data and/or the monitor signals received from the balloon-borne instrumentation an eight channel digital to analog converter unit was constructed. It was designed to be compatible with the IEEE-488 bus.

Its task was to receive an eight bit digital data word from the bus controller, to perform digital to analog conversion and to display the resulting analog signal on a designated microammeter. Each of the analog signals was also available on two output connectors for recording devices.

The D to A converter unit was designed to be a listener only. It could be addressed with the decimal codes 22 or 30 followed by one of the secondary codes 16 through 23. The secondary codes designated one of the eight D to A converters to be used for the conversion and the display of the digital data word.

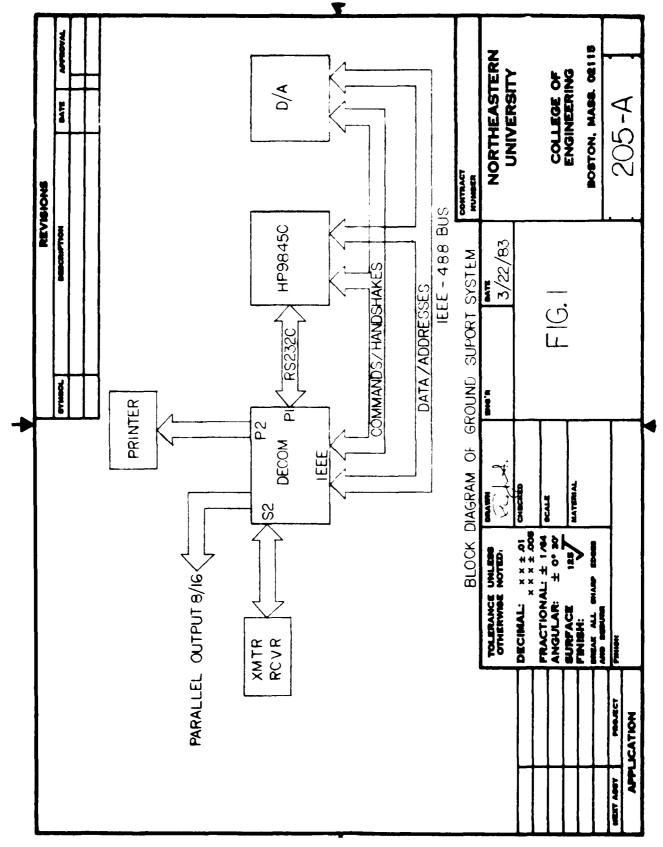
The circuit diagram for the D/A unit is shown in Figure 2. With the exception of the circuits directly interfacing with the IEEE-488 bus, standard CMOS devices were used to implement the necessary control logic. A TTL compatible high speed CMOS unit (U10) transmitted the data bus signals to the internal bus. Low power Schottky TTL gates (UIA,B) received the control signals and the low power HEXFETS (U2A,B) were used to perform the handshakes. To translate from the TTL levels of the IEEE-488 bus related devices to the CMOS levels, pull-up resistor network (Ull) was used. The inverters (U3A,B) and the gates (U9) detected the presence of the primary select code on the internal bus. If at the same time the ATN line was in the TRUE state, the output of U9B enabled the gate U7A. A transition of the DAV line to the TRUE state produced a positive pulse at the output Q2 of the monostable U5A, which through the gate U7A SET the flip-flop U6A. Thus the primary select code was detected and the gates U8 were set to receive the secondary select code. When one of the eight secondary codes appeared on the internal bus, gates U8 became enabled. Next,

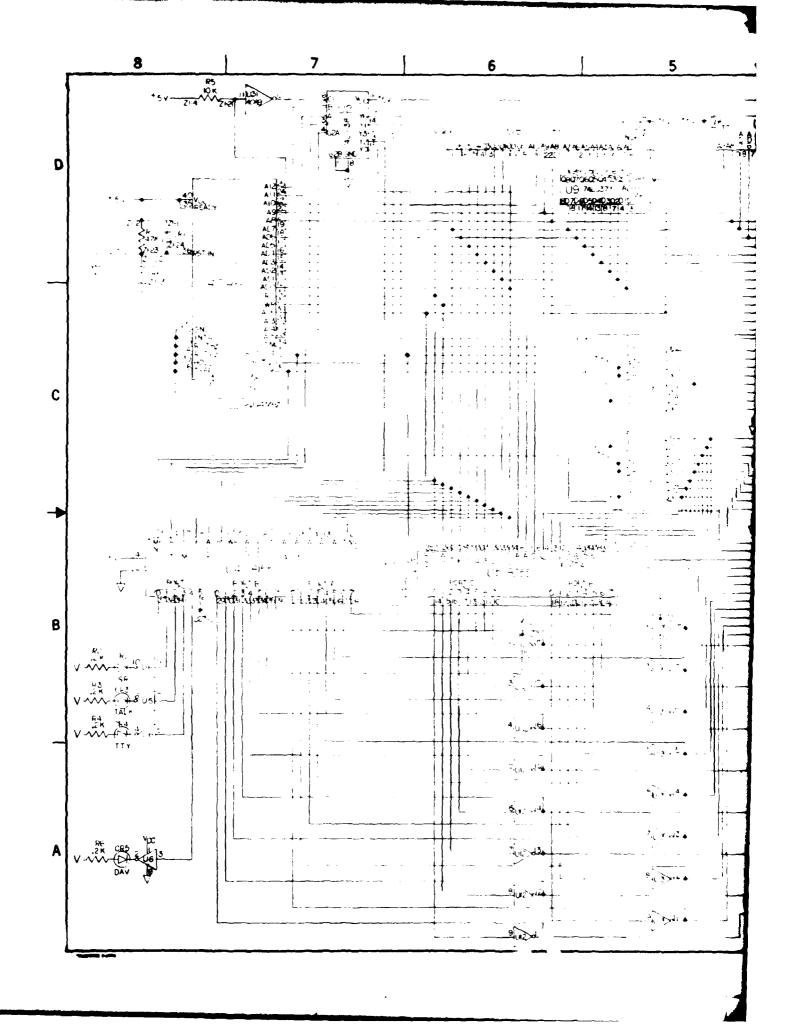
when the DAV line made the transition into the TRUE state, the four least significant bits of that code were latched into U12 by a pulse from U5A passing through the gate U8B. At the same time U4A was SET. Output \mathbf{Q}_2 of U4A enabled the gate U7B. When the ATN line returned to the FALSE state, signifying that the next byte on the bus was data, the output of U12 was enabled. The ATN signal was passed through U1A, U3C, U7B and U3E. Thus one of the D to A converters (U13-U20) was selected ($\overline{\text{CS}}$ low) by U12 responding to the previously latched secondary code. The digital data word present at that time on the internal bus was latched into the internal register of the selected D to A converter by a pulse originating from $\overline{\mathbf{Q}}_2$ of U5A. Once again that pulse was generated by the transition of the DAV signal from FALSE into the TRUE state. Since latching occurred on the trailing edge of the pulse, setting time for the data and the set-up time for the converter was provided.

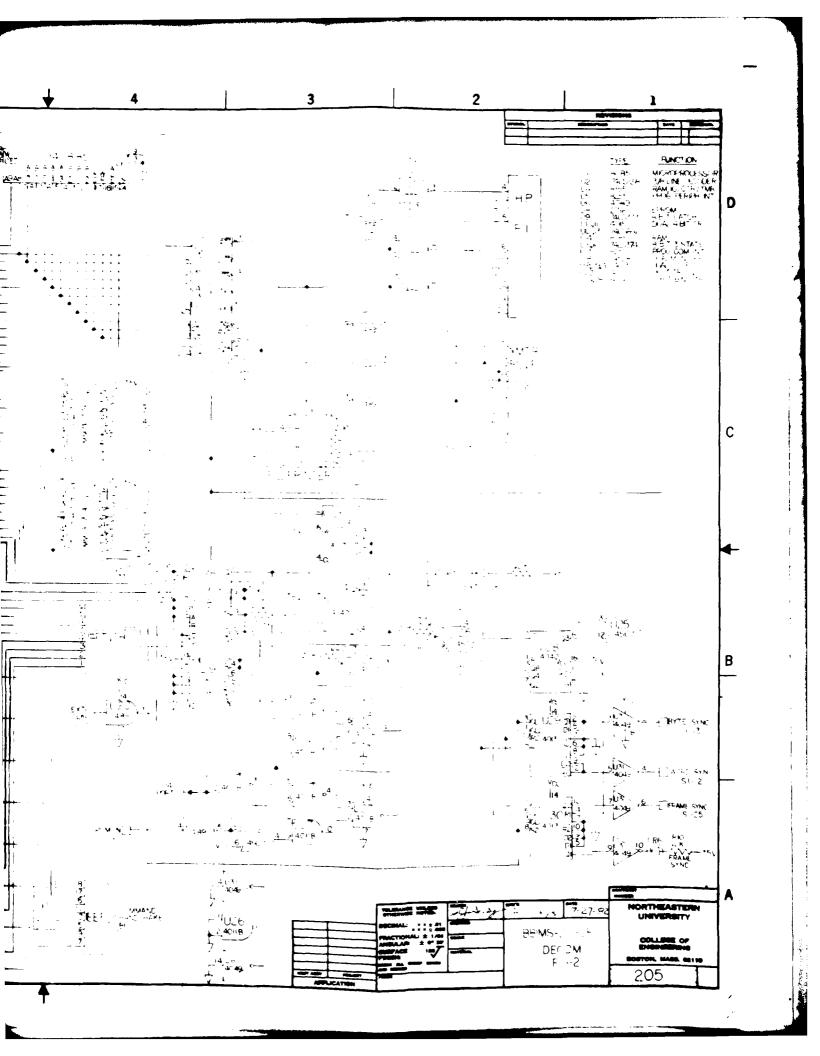
To indicate to the user of the unit, that data transfer into the D to A converter had taken place, the LED CR3 was lit whenever the outputs of the U12 were enabled. Flip-flop U6B generated the pulse to turn the LED on.

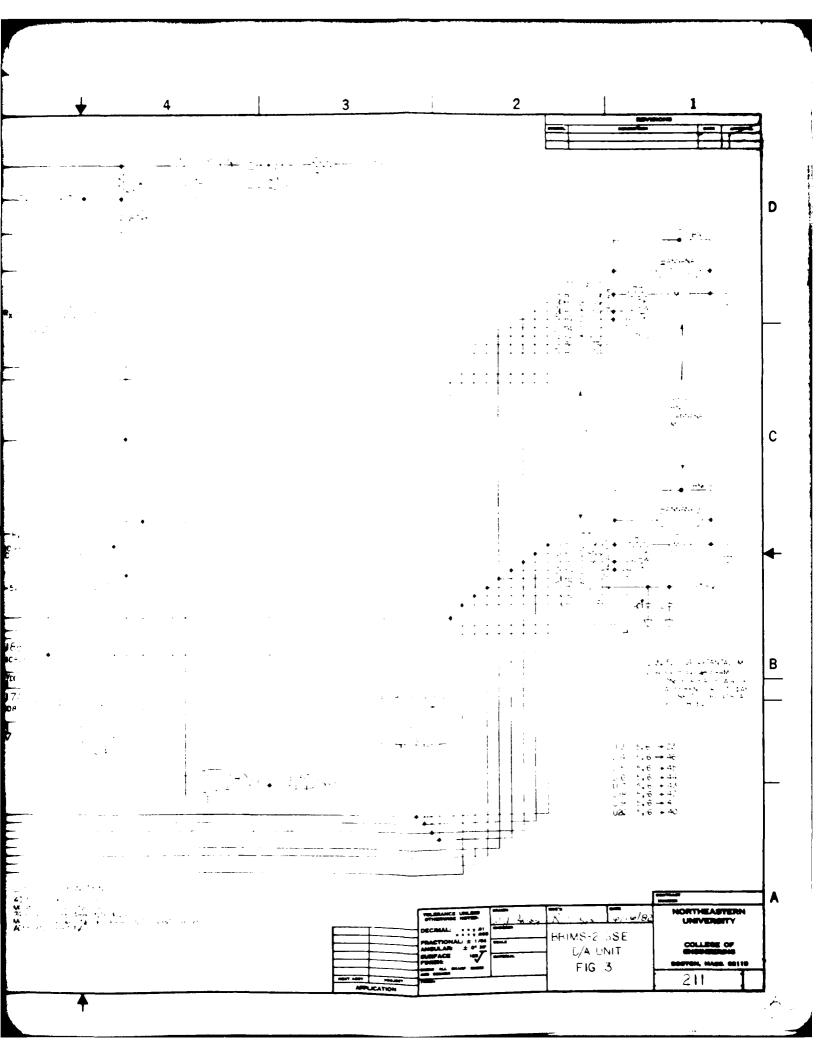
FALSE state on the ATN line disabled the code selection gates. Transitions of the DAV line from TRUE to the FALSE state with the ATN line in the FALSE state prepared the secondary code detection circuits for the next select code. The transitions generated pulses at U5B. These pulses passing through U7C clocked the flip-flops U6A and U4A into the reset states. The secondary selection code detection gates U8A became disabled. Also, the output of U12 returned to the ONE state. Thus the A to D converters were deselected.

The handshake signals were generated by U4B. A pulse from $\overline{\mathbb{Q}}_2$ of U5A produced the NRFD and $\overline{\text{NDAC}}$ outputs. A pulse from $\overline{\mathbb{Q}}_1$ of U5B at the trailing edge of the DAV signal returned the handshake lines to $\overline{\text{NRFD}}$ and NDAC states. Initialization pulse for the control circuits at power turn-on was generated by U3D.









APPENDIX - BBIMS-2 DECOM SUBROUTINES

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DECOMMUTATOR SUBROUTINES

XFR

This subroutine is entered upon the RST 7.5 interrupt. It transfers the present PCM data word into a temporary storage. The contents of all registers are saved in the stack. The appropriate address for the DCM routine is established and then transferred into the program counter.

DCM

This routine controls the decommutation by calling upon appropriate subroutines to process the words of the PCM data frame. To accommodate other PCM data frame formats the routine must be changed.

BK

The registers are reloaded from the stack. Interrupts are enabled and the system is returned to the interrupted task. This routine is entered at the end of every subroutine called by DCM.

WRD

When valid data is present in the PCM data frame, a selected word is stored in the buffer RAM. The storage locations within the buffer and the stored word counter are controlled by the routine.

WRD 1

This routine determines if new data and/or communications from the BBIMS flight control unit are present in the PCM data frame. It sets appropriate flags (DV and TTY) and lights the monitor LED's.

TTY

Communications from the BBIMS flight controller are stored into a temporary buffer within the RAM. When USART is empty a stored character is transferred into the USART for transmission to the computer.

SFID

When the subframe identification word is a binary 7ERO, the contents of the subframe word counter are transferred into the first location of the subframe buffer. Pointers for the new storage buffer and the full buffer to be read out are established.

SFWD

This routine stores the subframe words and keeps count of the number of words stored.

FS1

Checks for the first frame sync word. If the word is not found it sets an error flag. When DV flag is detected the routine closes the present minor frame storage buffer by placing the stored word count into the first buffer location. New storage buffer address and pointer are established. In the case that the new storage buffer is in the process of being read out, the read pointer is pushed to the next buffer. Also the turn-offs of the monitor LED's are controlled by this routine.

FS2

Outputs synchronization pulse to the PCM, data bit counter. Checks for the second frame sync word and re-establishes frame sync when necessary.

SEARCH

On turn-on or reset the search routine is entered as a part of an initialization process. It establishes the polarity of the incoming PCM data and the original frame sync for the decom system. From SEARCH the program enters IDLE.

IDLE

In this routine the system searches for full storage buffers. The minor frame data buffers have the priority. Only when all of these are empty the subframe data buffers are checked.

E488

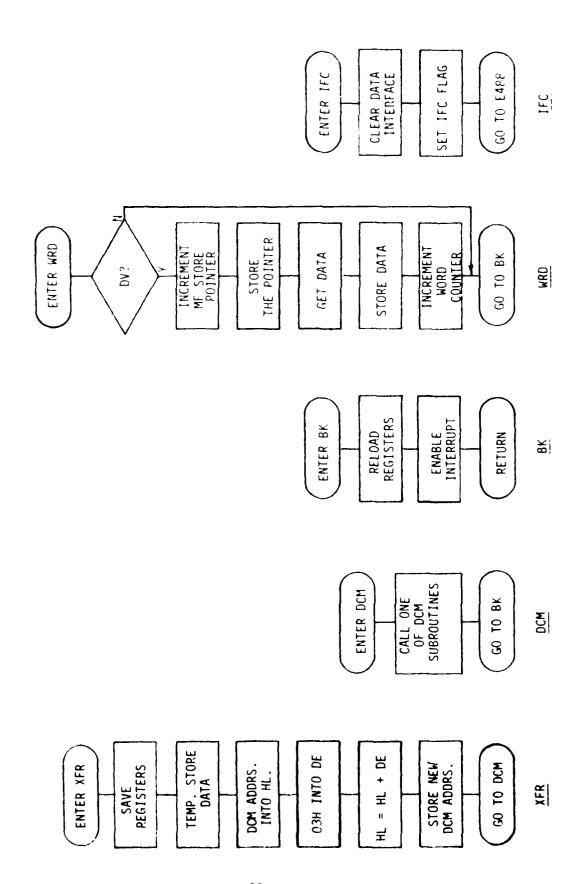
This routine requests service and establishes the first contact with the computer before data transmission. Parallel polling ID method is used.

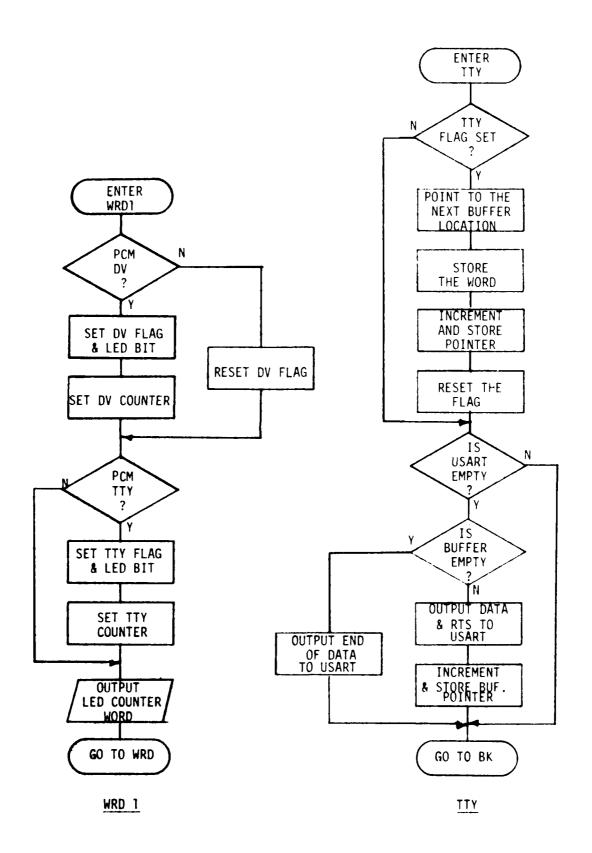
OUTPUT

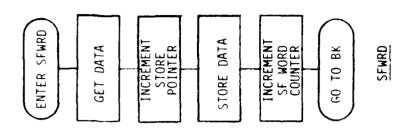
Performs the necessary handshakes and transfers data from the storage buffers of the decom to the computer. The first data word identifies the data as coming from a subframe or a minor frame buffer. Data transfer ends with the EOI control code.

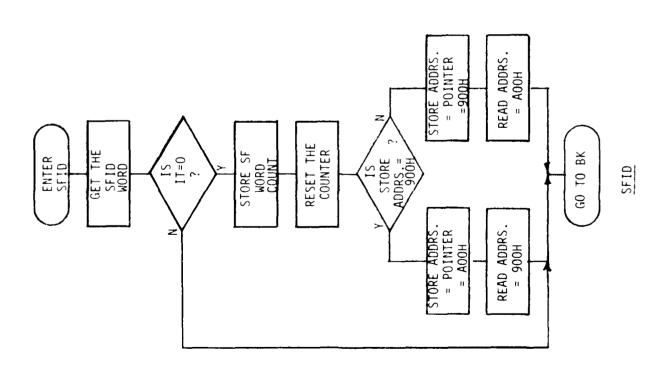
IFC

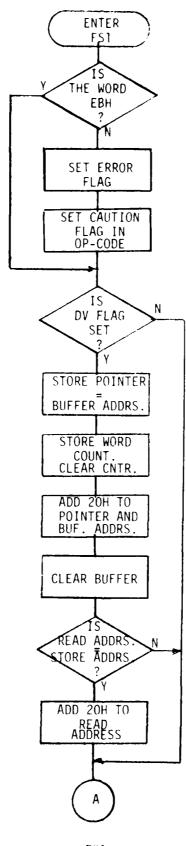
Clears the data interface upon detection of the IFC command during the data transmission. E488 is entered from this routine.

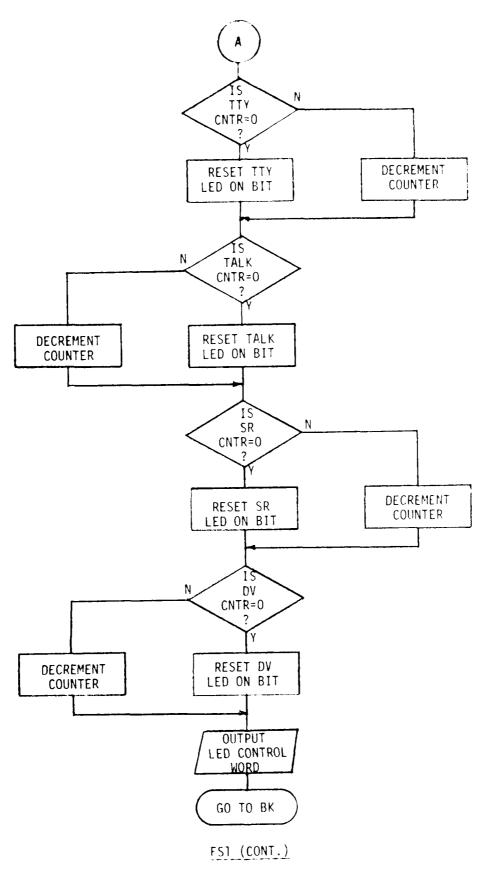


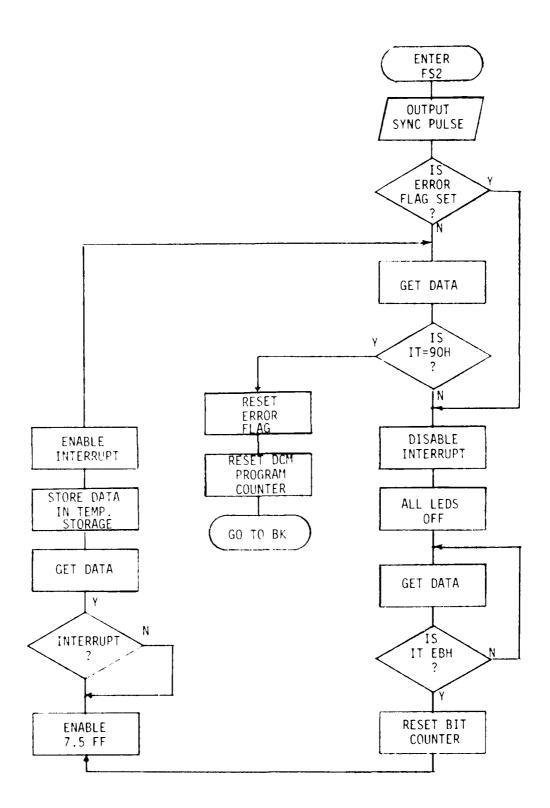




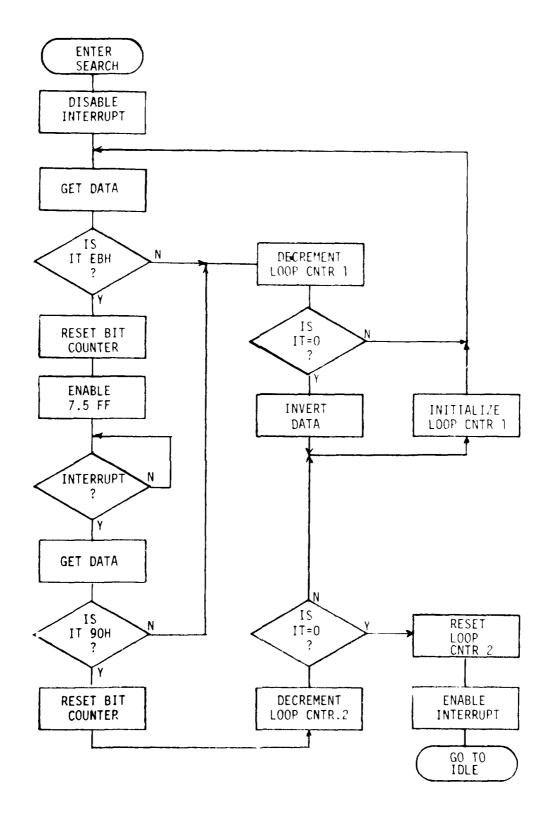




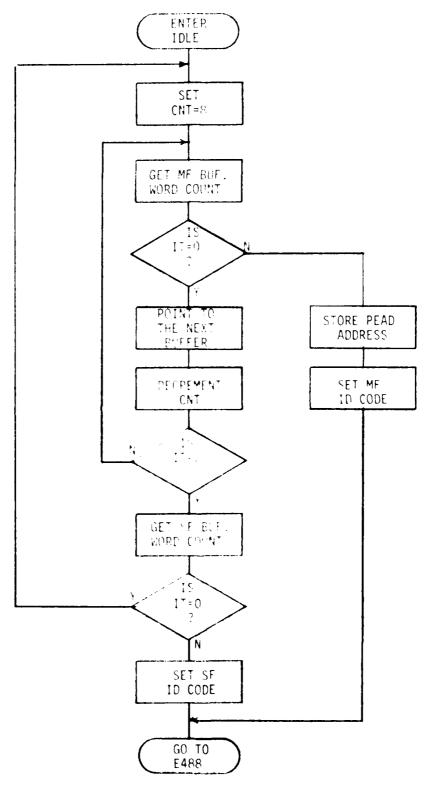




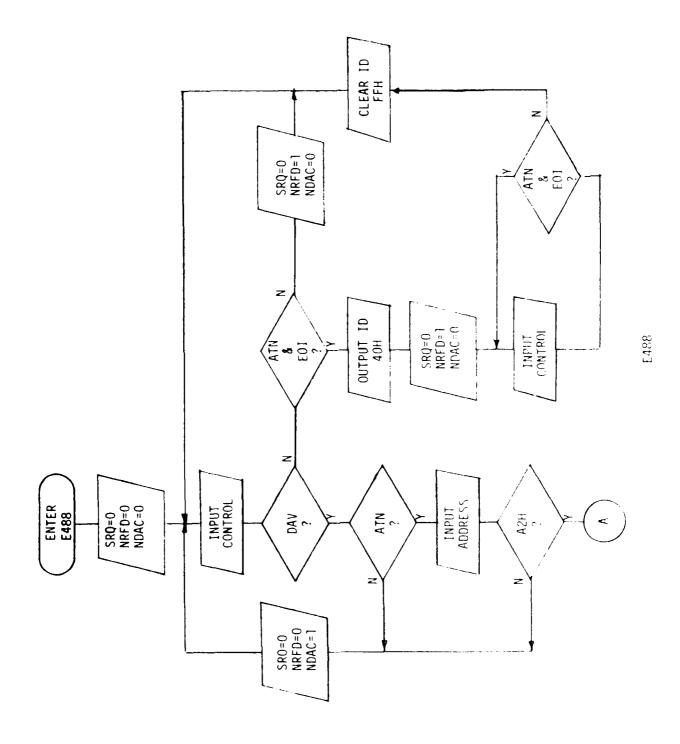
FS2



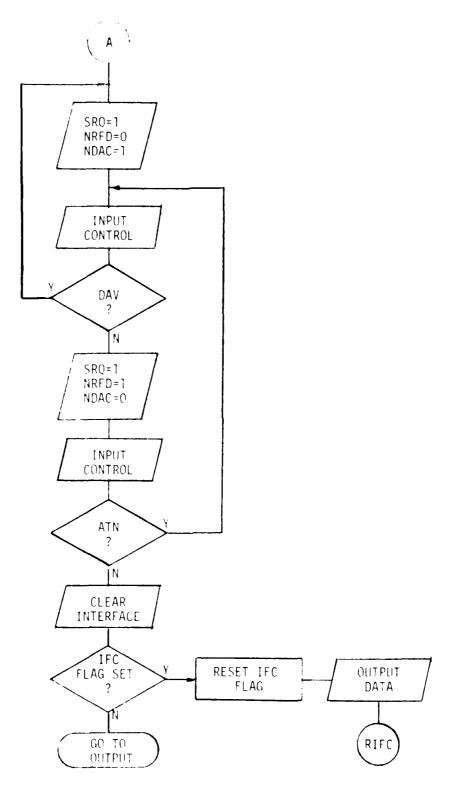
SEARCH



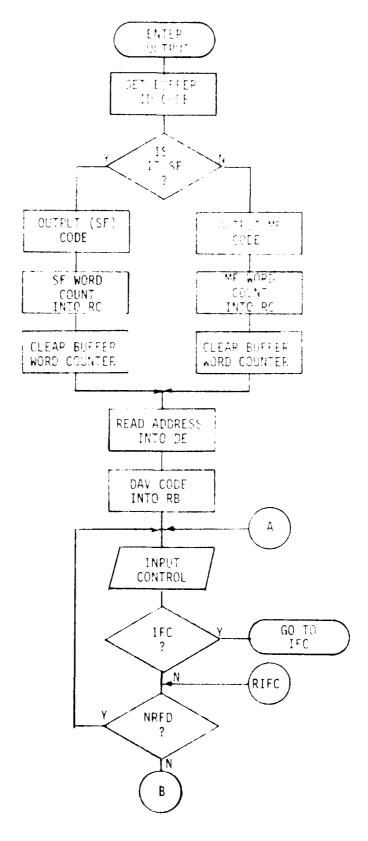
IDLE



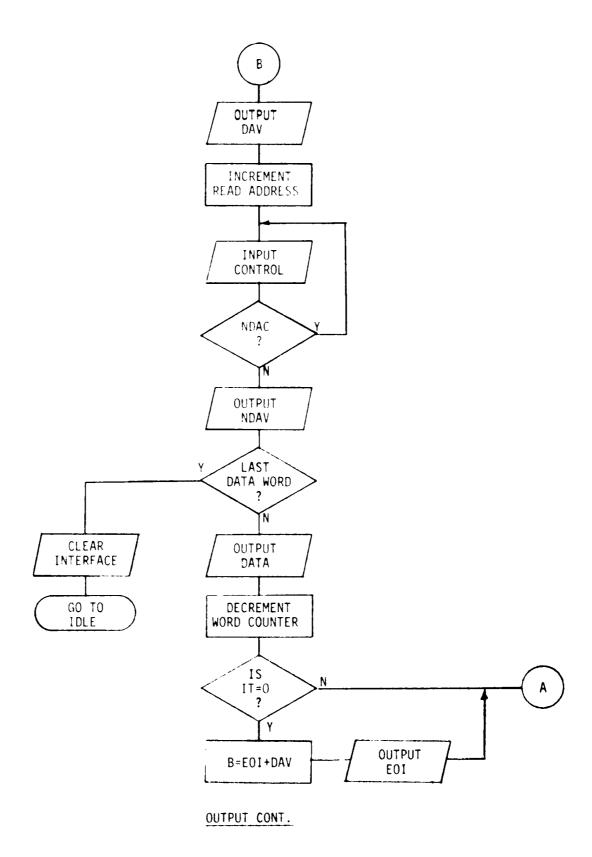
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E488 CONT.



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V. 112 W.

A list of the engineers who contribute to the production given below:

J. Spencer Rochefort, Protessor of Electrical Engineering and Principal Investigator.

Raimundas Sukvs, Senior Research Associate, Company

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